

Performance Simulation
of
Tethered Rotor-Blade
Lifting Vehicles

Airborne Wind Energy Conference
September 29, 2010

Presentation Objectives:

- *for the 'Rotor Naïve'*
- *for the 'Rotor Aware'*
- *for the 'Rotor-head'*

Subjects to be Addressed

- Applying Rotors to AWE as a Lift Element
- Modeling Tools Required
- Introduction to Rotor Idiosyncracies
- Modeling Rotor Lift
- Performance Simulation Example

Harvesting Wind Energy with Rotating Blade Lift Elements

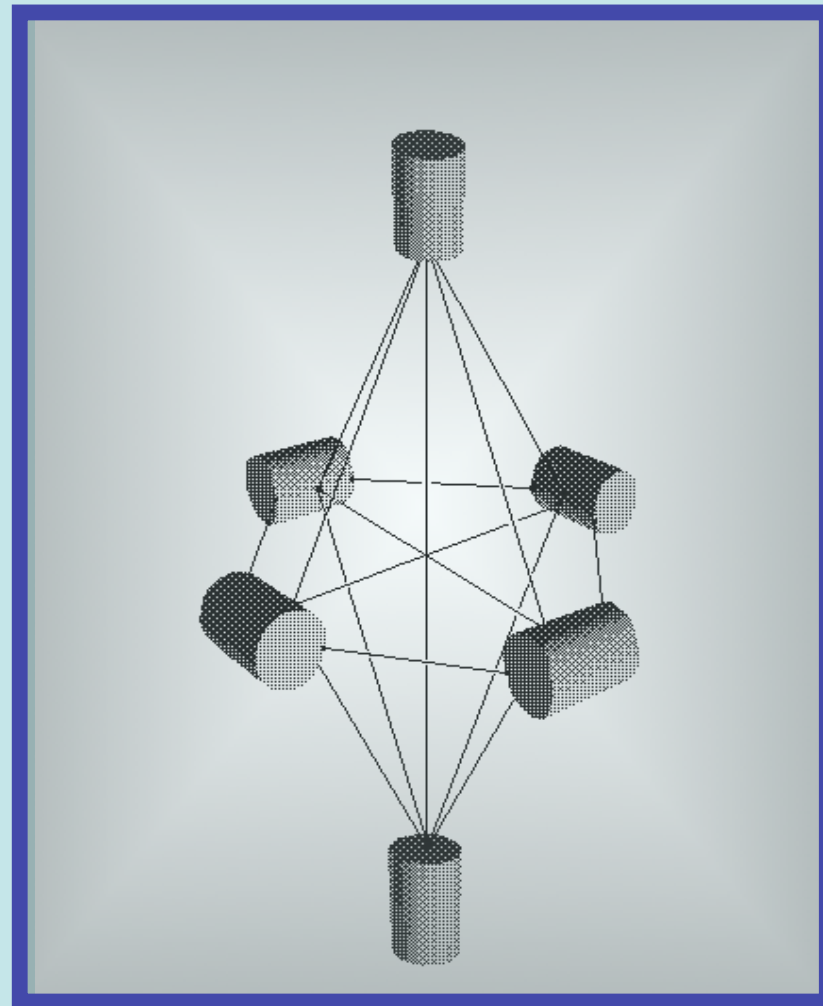
- Pull Hard on a String - “Ground-Gen”
- Carry a Generator Aloft - “Air-Gen”
- Use Excess Autorotation Torque - “Air-Gen”
- Hybrid - Air-Gens using Turbines as multi-purpose

Modeling Tool

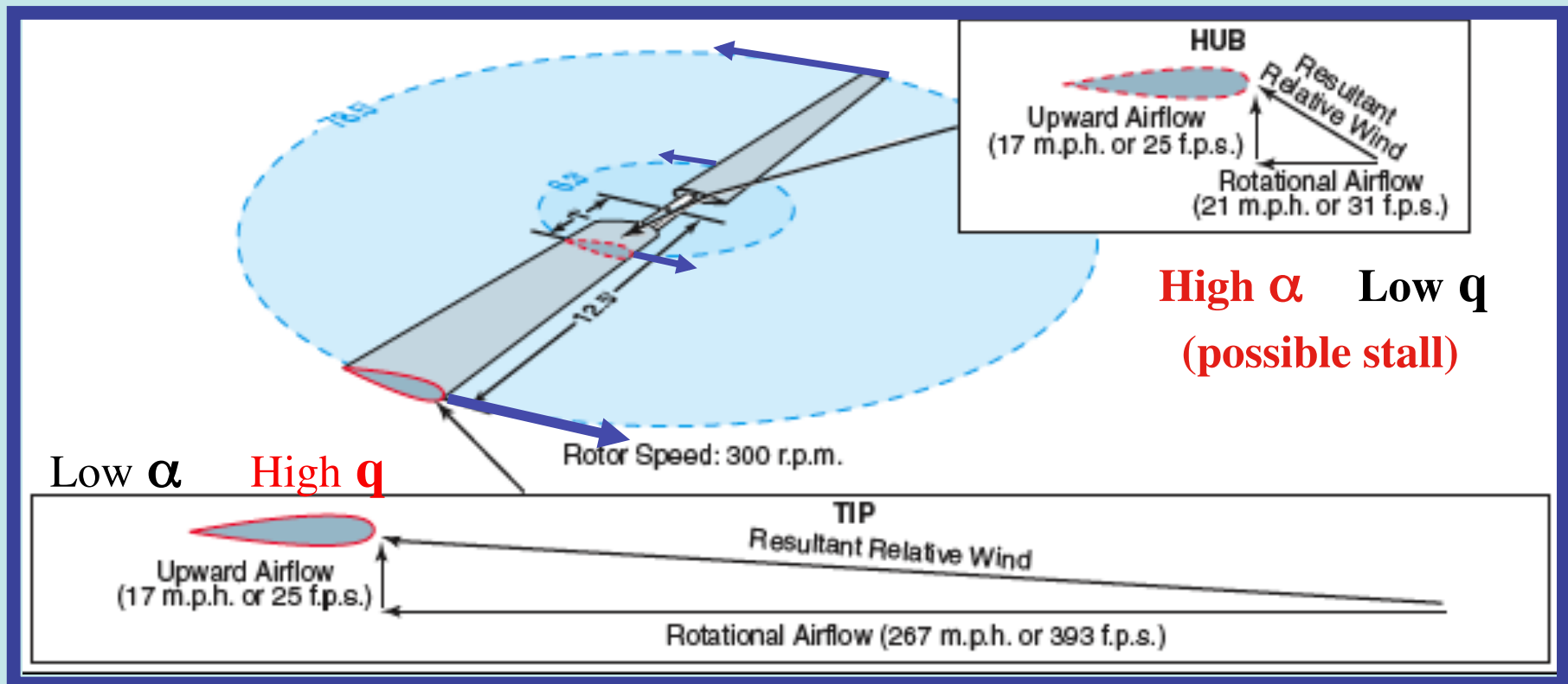


- **Point-Synthesis Finite Tether models, providing:**
 - Free or Constrained*
 - Tether Deployment-dynamics*
 - Non-Uniform Tethers*
 - Aero-dynamics*
 - Thermo-dynamics*
 - Electro-dynamics*
- **Wind and Gust Atmospheric**
- **Industry Vetted**
- **User-friendly F99 code & Docs**
- **Multiple 6 DOF Bodies w/Arbitrary Tether Connectivity**

Example of General Connectivity



What a Rotor Blade Sees



Autogyro - vs - Helicopter

- Inflow is From Opposite Direction -

#1



Autogyro

Helicopter

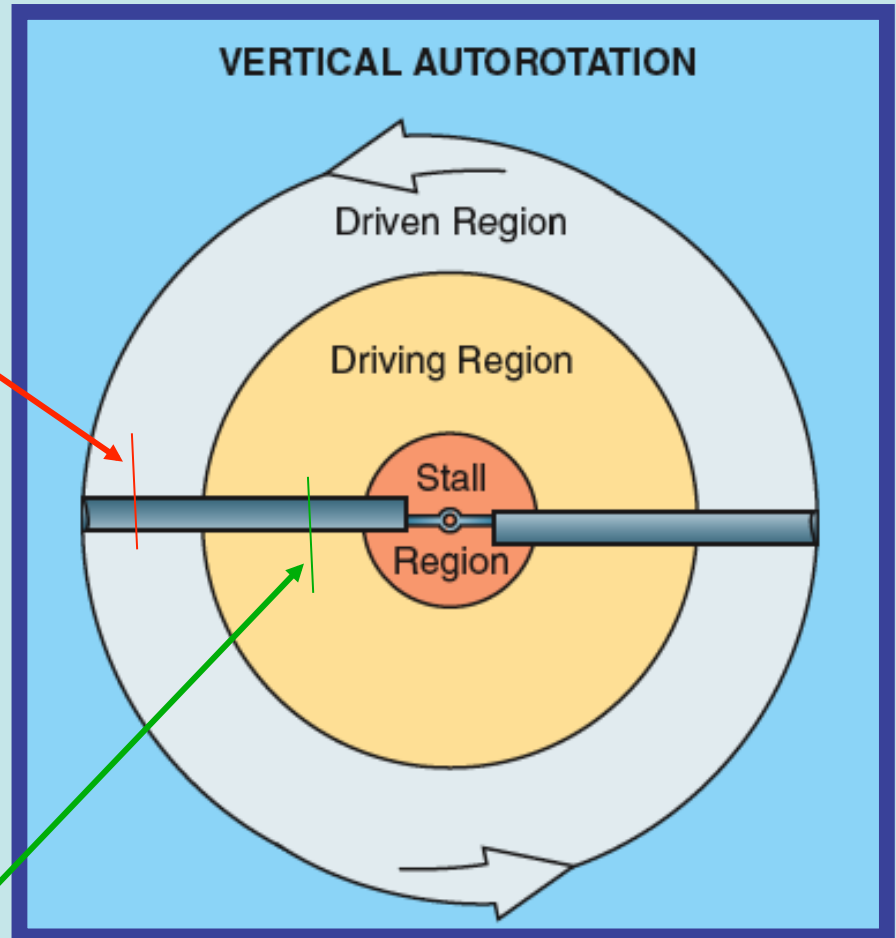
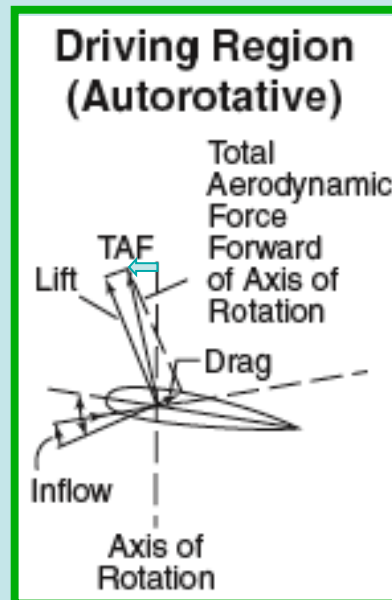
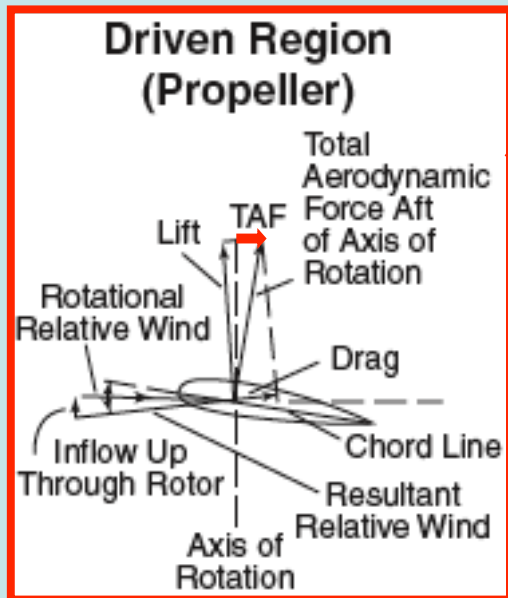
#2

- *Autogyro Blades Power their own Rotation*

-VS-

- *Helicopter Must Supply Power for Rotation*

How Does Autorotation Occur



Putting Auto-Rotation to Work

- What Happens when Autogyro is Constrained by a Tether?
 - Changes how Autogyro functions compared to Free-flight
 - Can Experience *Large Impinging Relative Wind Velocities*
 - *Steady-State* Disk-Loading can **Exceed** *Normal Free Flight*

 RPM is King! 

The Impact of Relative Wind

- Wind **Powers** the Autorotation, thus, LIFT
- Wind can produce **Stall regions** in the Rotor
- Creates Complex Blade Aerodynamic Environ.
 - function of Impinging Wind Level and Direction
 - function of RPM
 - function of Radial Position along a Blade
 - function of a Blade's Azimuth Position

Rotor-Map Schemes

- **Pros:** Reduces High CPU Cost
 - can be useful for “Real Time” applications
 - can minimize “Development” execution run-times †
- **Cons:** Labor-Intensive Up Front Costs
 - Unwieldy for Blade Configuration Changes
 - Complexity as Function of Independent Parameters
 - Easy to Exceed “Envelope” in AWE Simulation
- **Can be Problematic for AWE Exploration**



An Observation:

Developed via Blade Element Simulations

Blade Element Model

- Element-by-Element Modeling (*for each Blade*):
 - *Incident Flow* = $f(\text{radius}, \text{azimuth}, \text{RPM}, \text{incident wind})$
 - *Induced Flow* = $f(\text{airload}, \text{radius}, \text{azimuth})$
 - Specific 2-D Airfoil section = $f(\text{radius})$
 - Chord-width = $f(\text{radius})$
 - Blade Differential-Twist = $f(\text{radius})$
 - Blade “Collective” and “Cyclic” (*control parameters*)
- 2-D Aero Coefficients table-driven -vs- AOA & Mach
 - Requires wide-range aero data (“C81 tables”)

Inflow “Corrections”

- Why Needed
- Dynamic -vs- Quasi-Steady Induced-flow Models
- “QUASI-STEADY” Models

- “Glauert Correction”

$$V_i = V_{i0} [1 + r' K \cos(\Psi_R)]$$

where: V_{i0} is aver. “momentum-theory velocity”

r' is x/R

K ranges from 0.5 - 1.6

- “DYNAMIC” Models
 - Numerous models have been developed
 - Murakami & Houston - specific to Auto-rotation

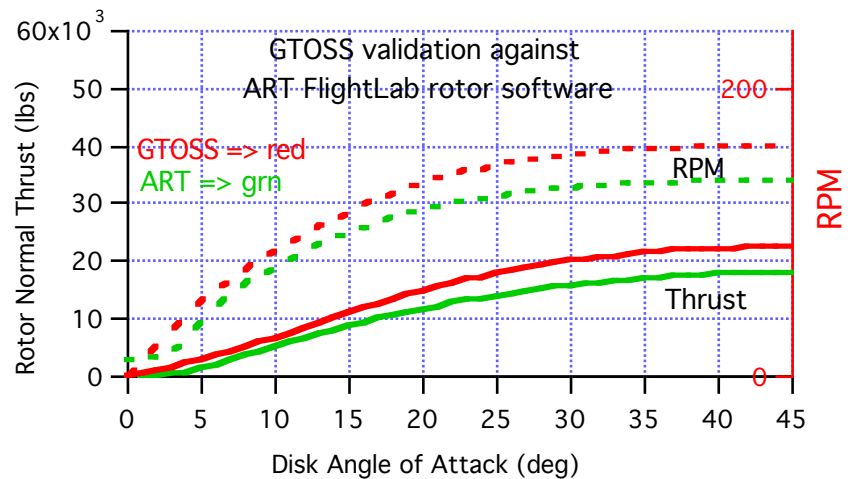
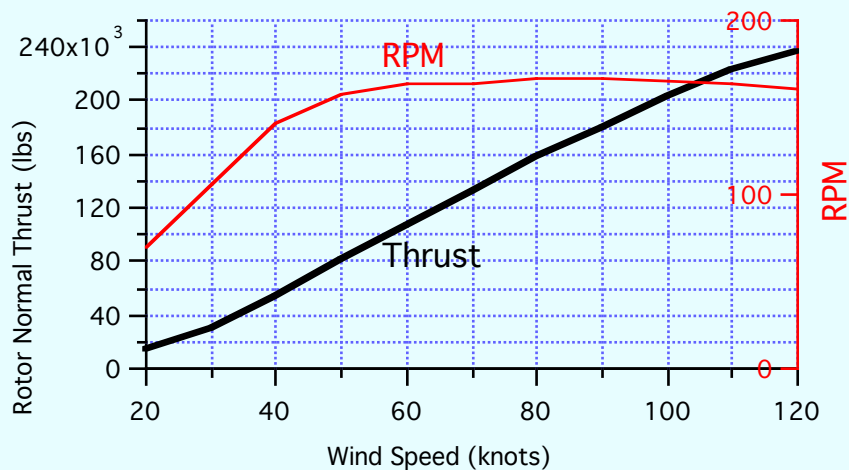
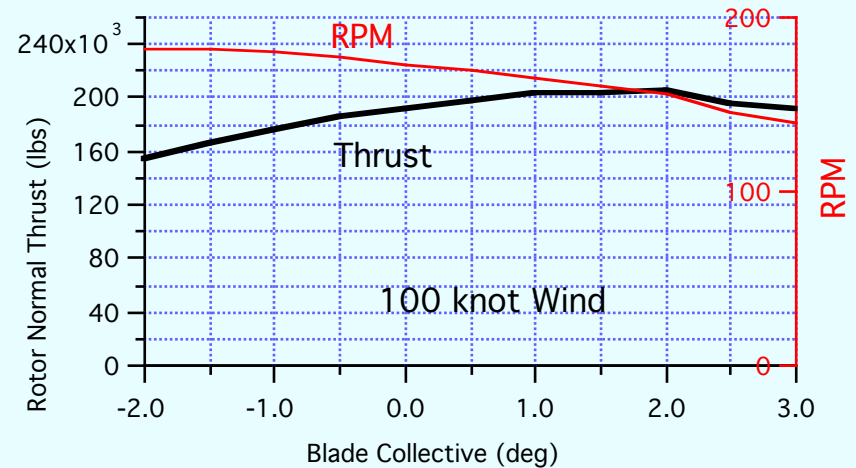
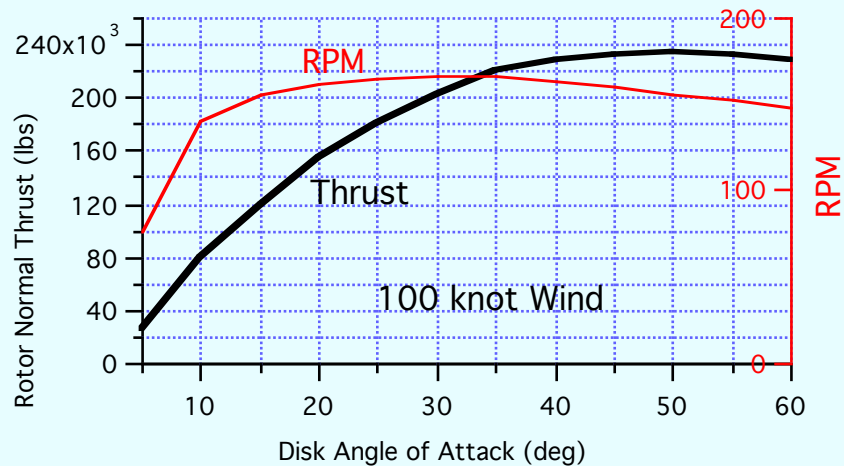
Tip Corrections

- Why Needed for the Blade Element Model
 - 2-D Aero data predicts Thrust
 - BUT**
 - Flow near Blade Tips reflect 3-D phenomenon
 - ⇒ Vortex Interactions
 - ⇒ Radial Flow
- *Complex* Corrections by Prandtl (and others)
- *Simple* Correction: if $r/R > K_{tip}$ then, **Lift = 0**
 - Sissingh sets the constant $K_{tip} = .93$

Example Rotor Characterizations

4 blade 100' dia rotor, NACA 0012 airfoil, 6' chord, no Twist

NOMINAL SETTINGS: 1 deg Collective, 100 kt Wind @ Sea level

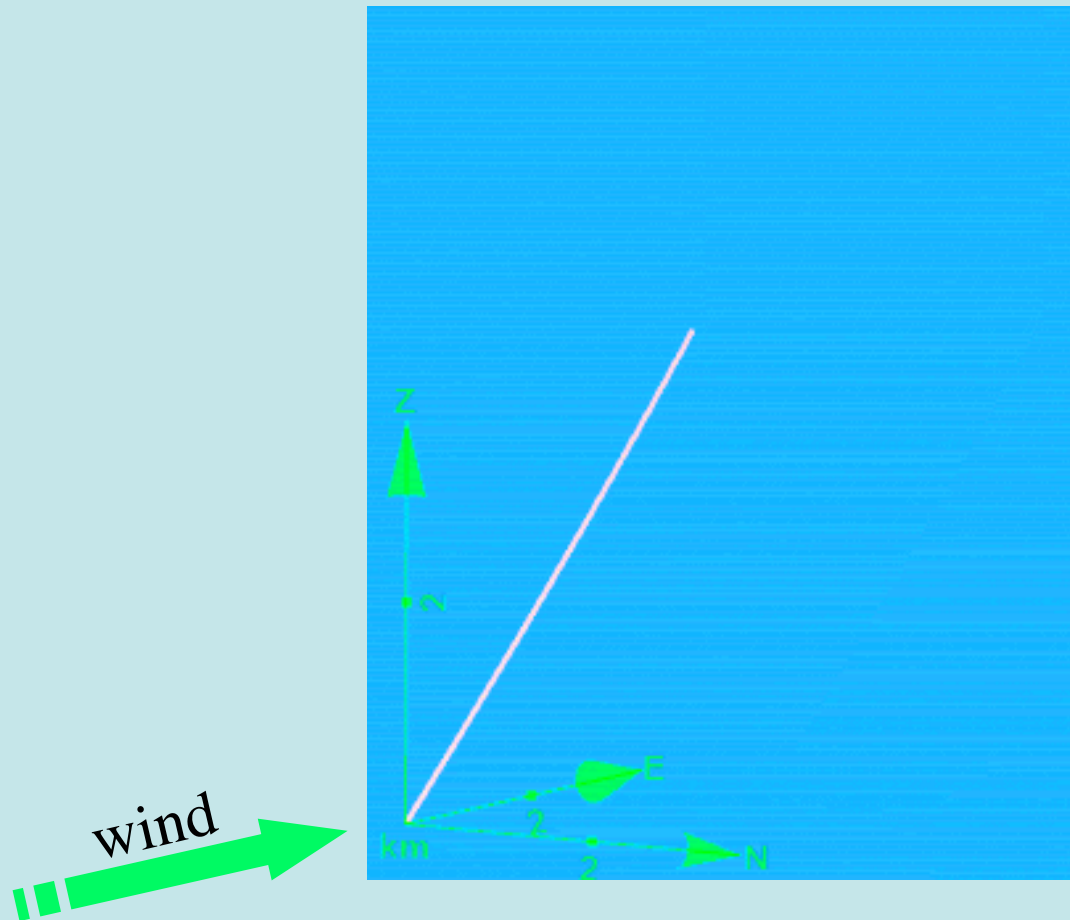


Movie of Typical Power Cycle

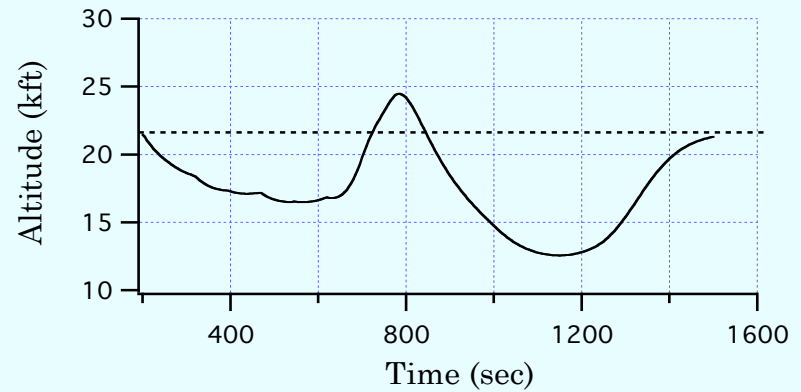
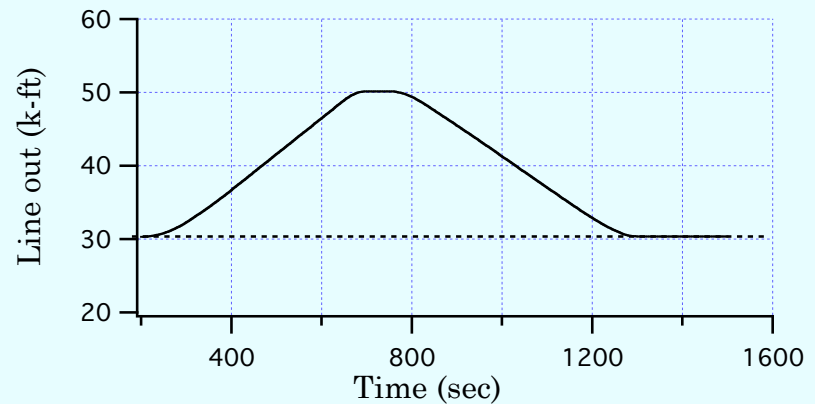
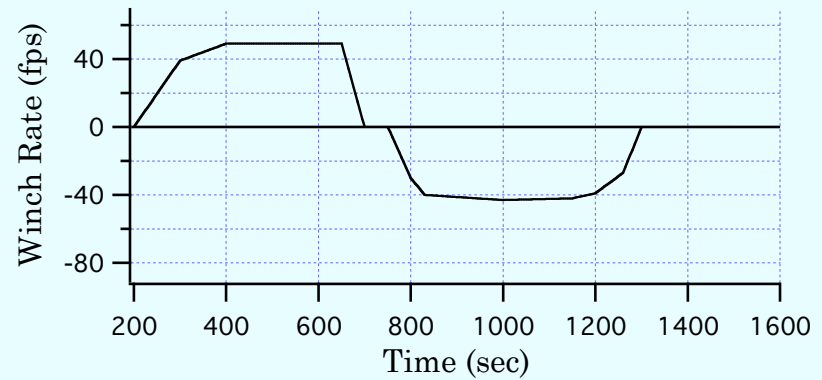
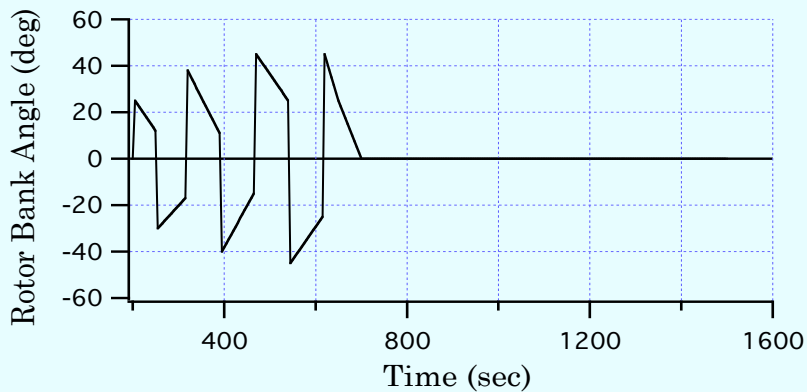
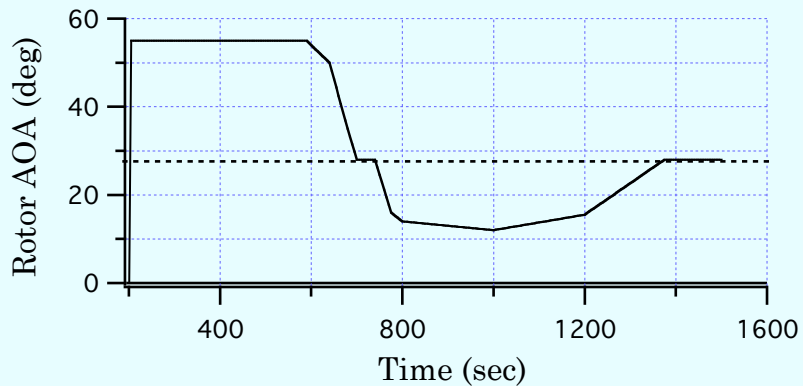
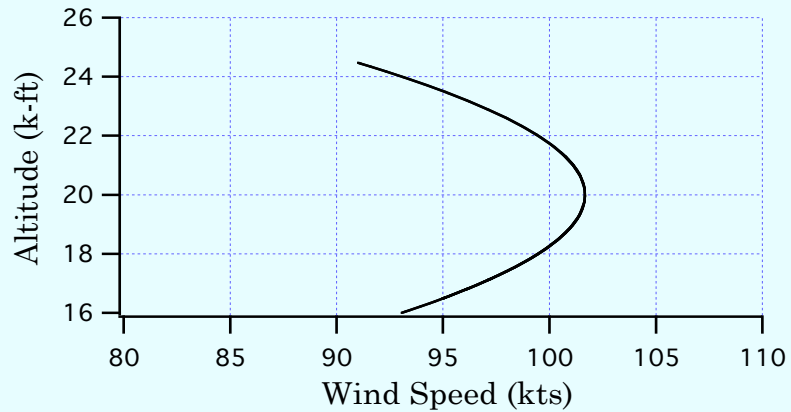
Tether Legend

Red = Highest tension

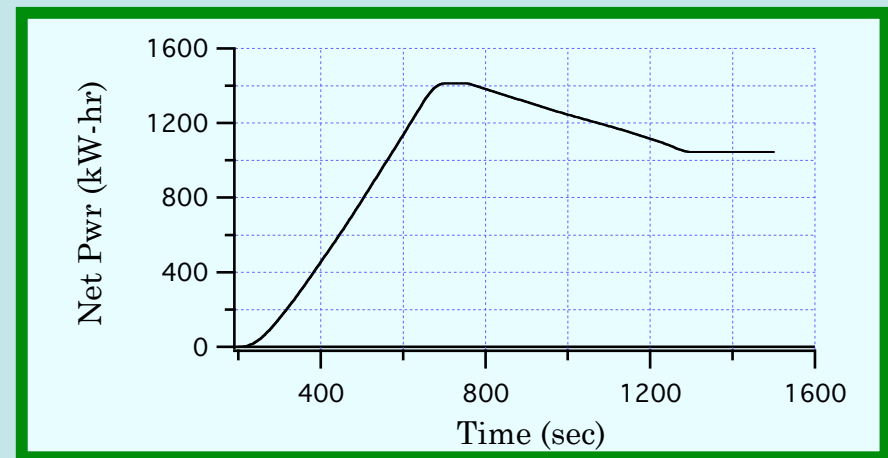
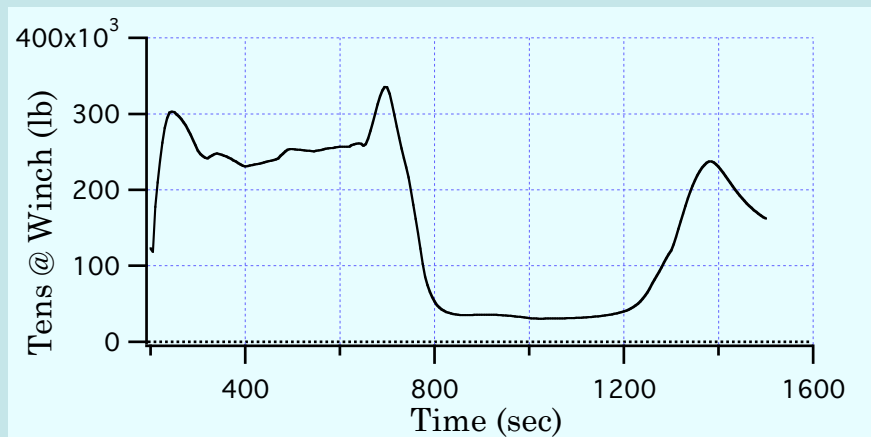
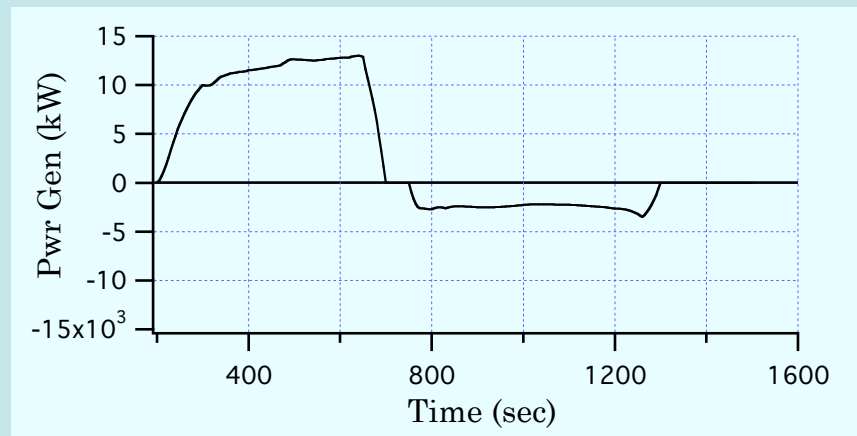
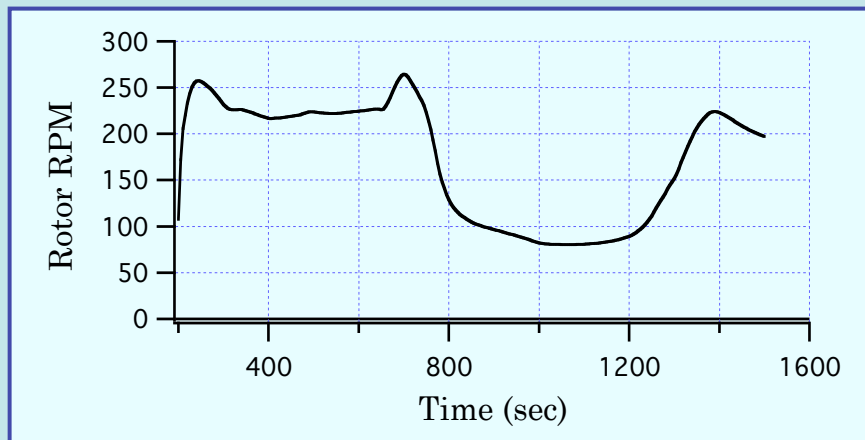
White = Lowest tension



Example Trajectory Kinematics



Example Trajectory Performance



**Thanks for your
Attention**